



Technology Advancement for Solar Sail Propulsion (SSP) for NASA Science Missions to the Inner Solar System

**A Summary Status Briefing to the
Solar Sail Technology & Applications Conference**

September 28, 2004

**Edward E. (Sandy) Montgomery
Technology Area Manager, SSP
In Space Propulsion Technology Project
NASA Marshall Space Flight Center**

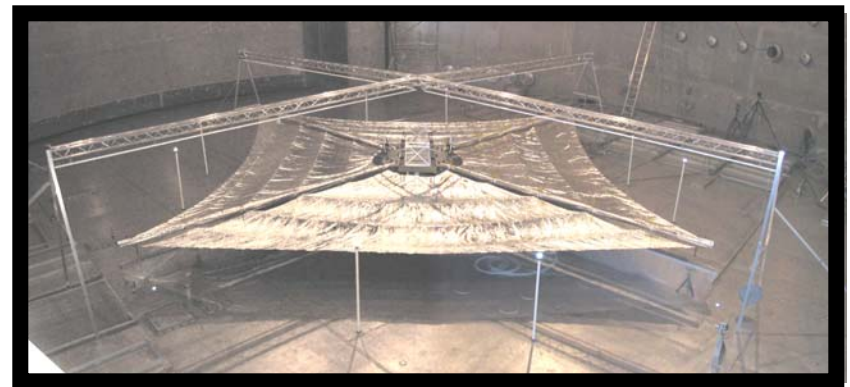


Contents



[Solar Sail Propulsion - SSP]

- Tech development roadmap
- Tasks funded to date
- TRL 6 Definition
- Future development plans





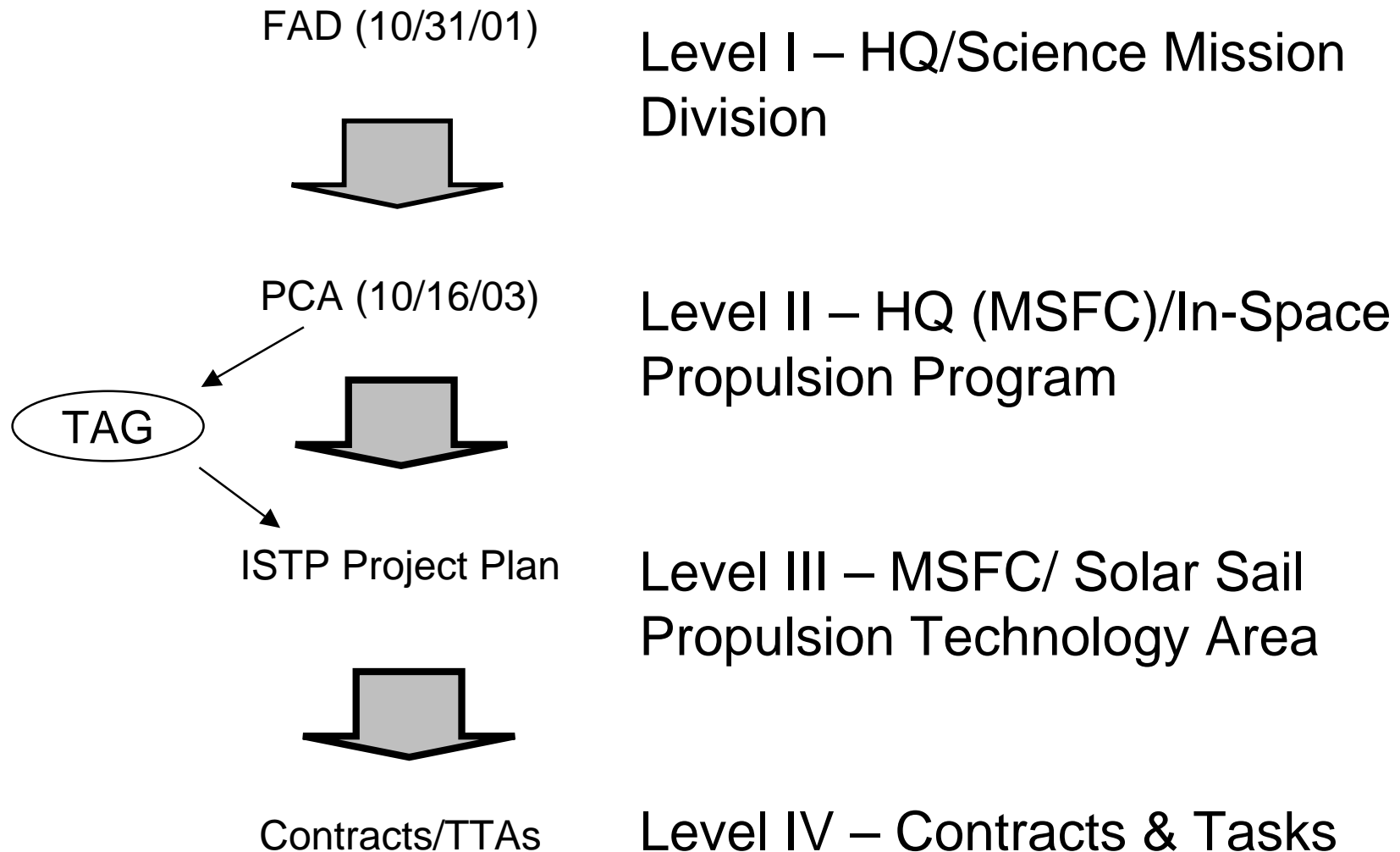
SSP Roadmap Foundations



- ◆ **Requirements Flowdown**
- ◆ **SSP mission statement**
- ◆ **mission pull**
- ◆ **Technology Advisory Group**
- ◆ **ROSS NRA**



Requirements Flowdown





SSP Roadmap

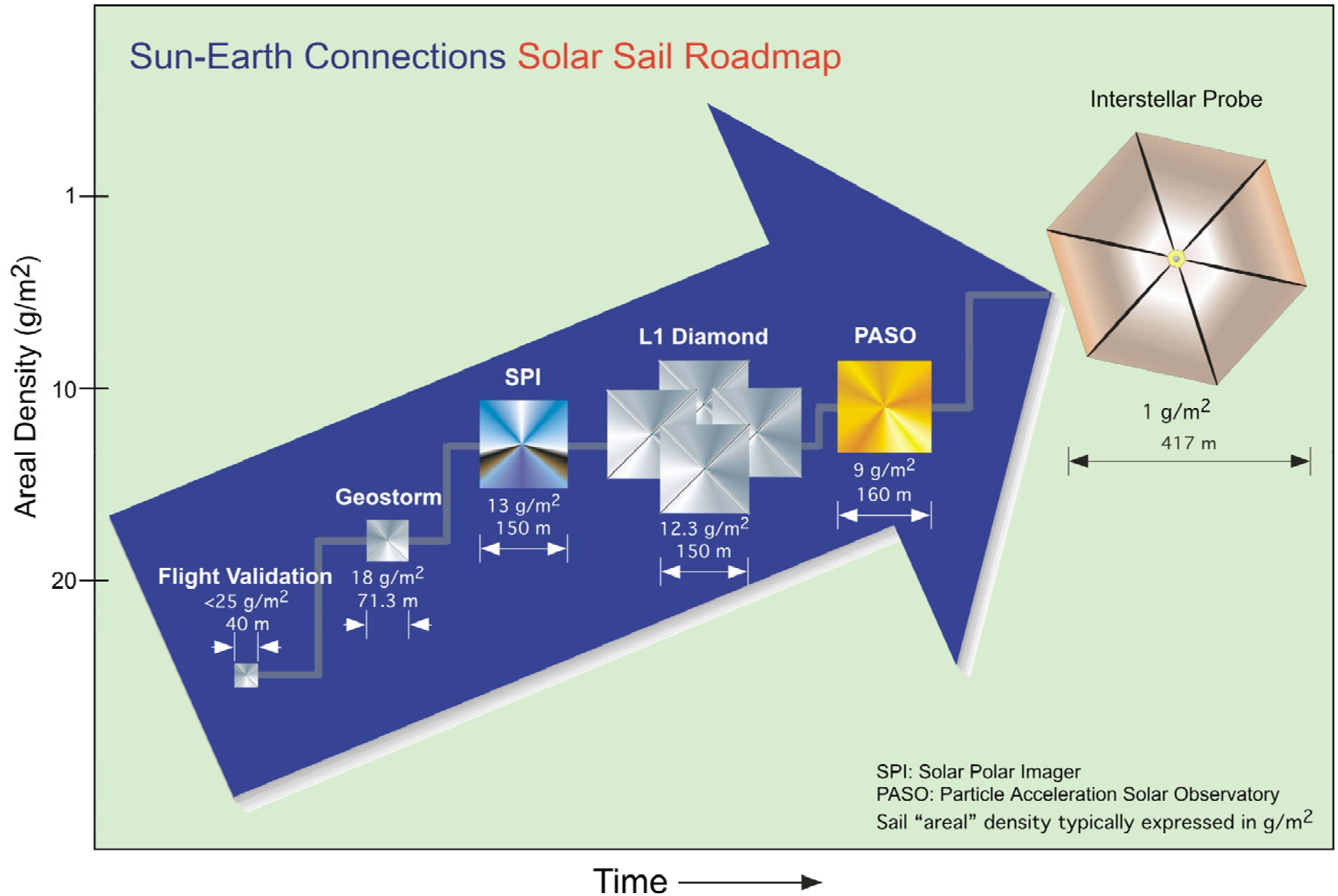


SSP Mission Statement (Draft Revision)

1. Lower the risk of solar sail technology as a primary propulsion system to a level enabling its selection for baseline on a NASA inner solar system science mission.
2. Solar sail propulsion demonstration and validation shall be scalable to inner solar system NASA missions
3. Attain closest approach to TRL 6 possible on the ground by the end of FY06
4. Maximize the potential for flight validation through partnerships and collaboration.

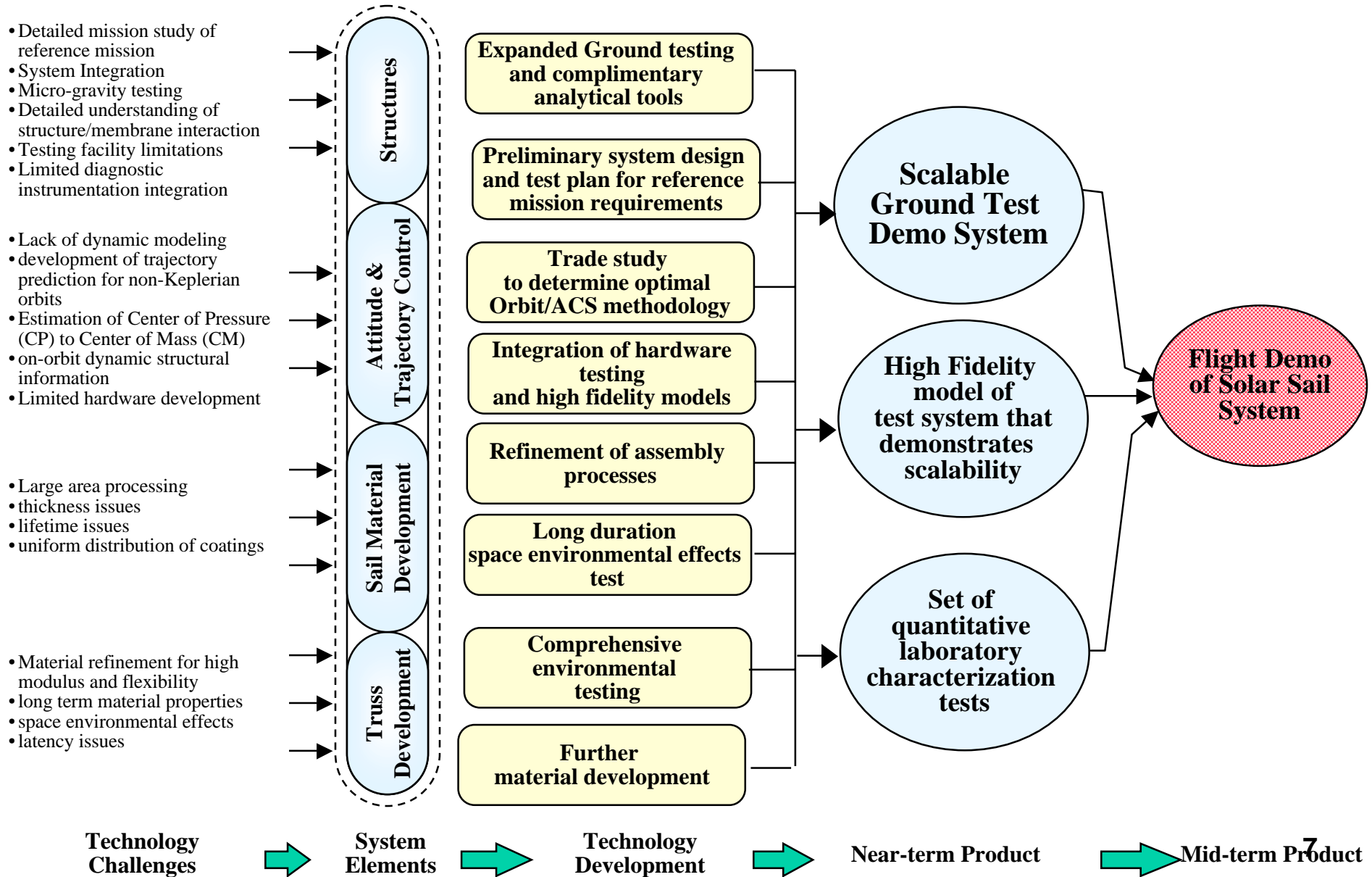


SSP Roadmap: Driven by Specific & Focused Mission Pull





Solar Sail Technology Assessment Group Roadmap In-Space Propulsion Program





Sail Technology Classes



Mission Class	Timeline	SOA	Technology Challenges	NASA Mission Application
GEO/GTO Short Life	Past/Now	Encounter(?), Cosmos, ST-7, Znamya, Inflatable Antenna Exp	AO, radiation belts effects, high GG torques	None
1 Au	Near Term	ISP Ground Demo, ST-5 Geostorm ST-9 SSFV	Validation in a space environment, Infusion into mission applications	L1 Diamond Solar Polar Imager (SPI)
<0.25 Au	Mid-Term	Mission analysis, Future ISP work	Materials environments, Thrust vector range, Lightweight system, 100s m system scale size	Particle Acceleration Solar Observatory (PASO) Titan Explorer Saturn Ring Observer
Extra Solar	Far-Term	Mission concept analysis, Gossamer, Encounter(?)	Ultra-lightweight system Integrated system architecture Sub to kilometer system scale size	Interstellar Probe (ISP) Geospace System Response Imagers (GSRI) Outer Heliosphere Radio Imager (OHRI)

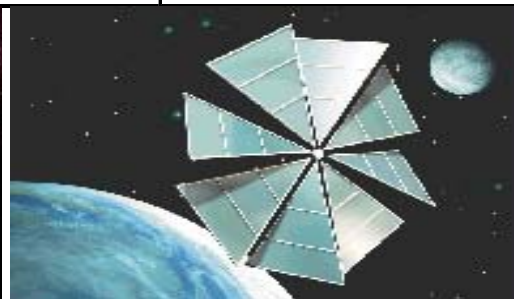
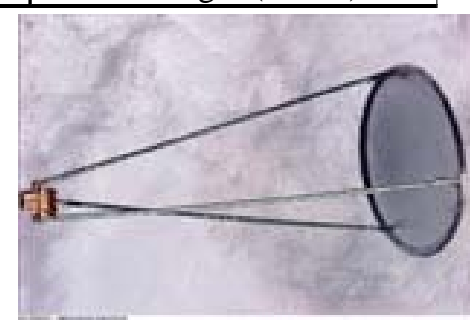
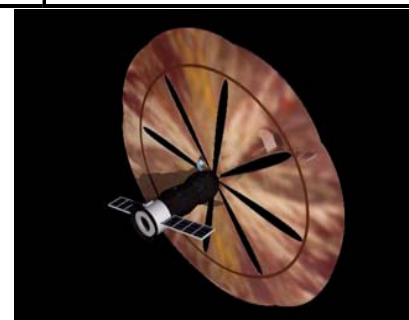
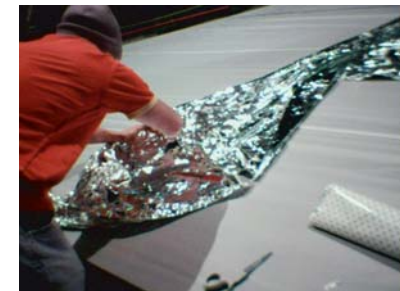
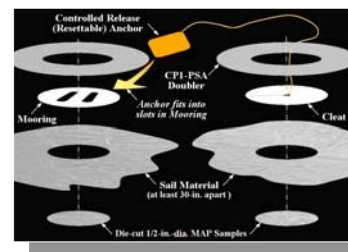
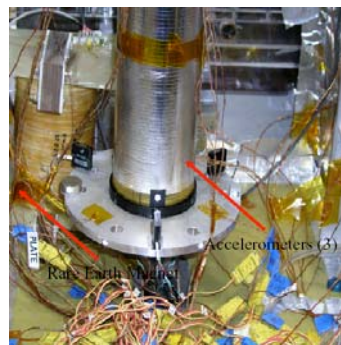
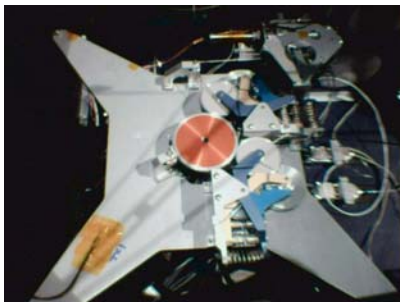
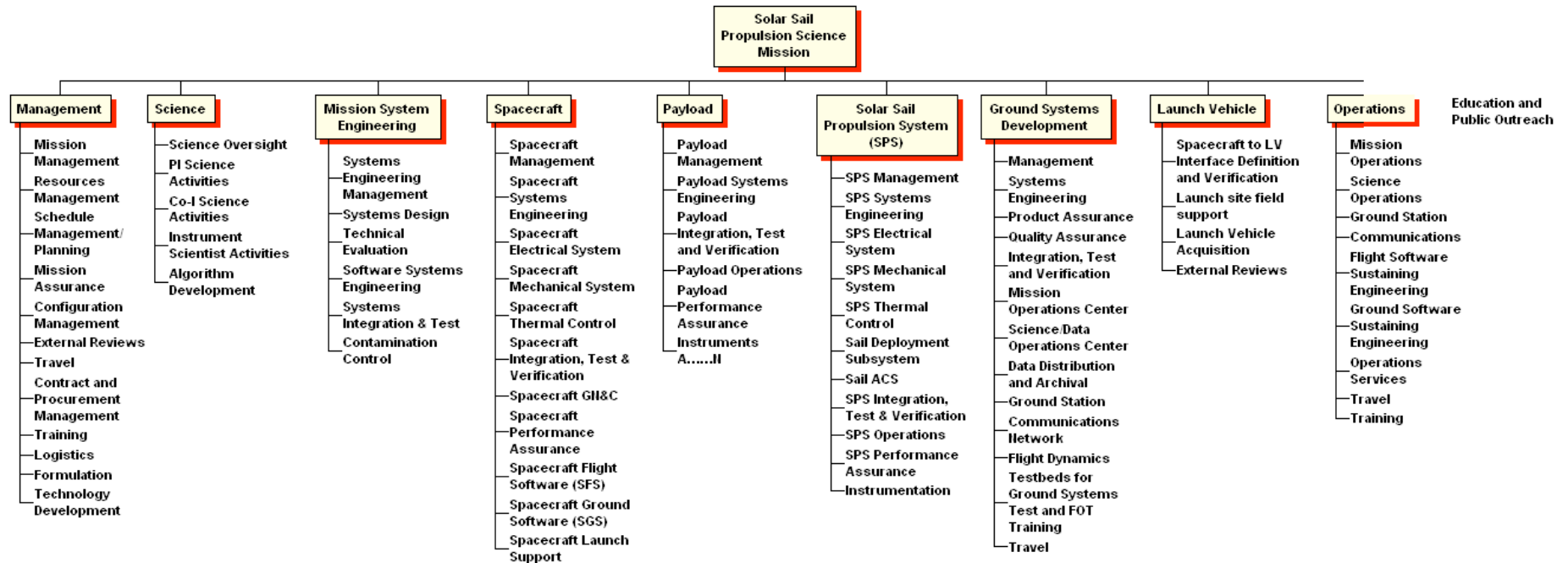


Photo courtesy of The Planetary Society



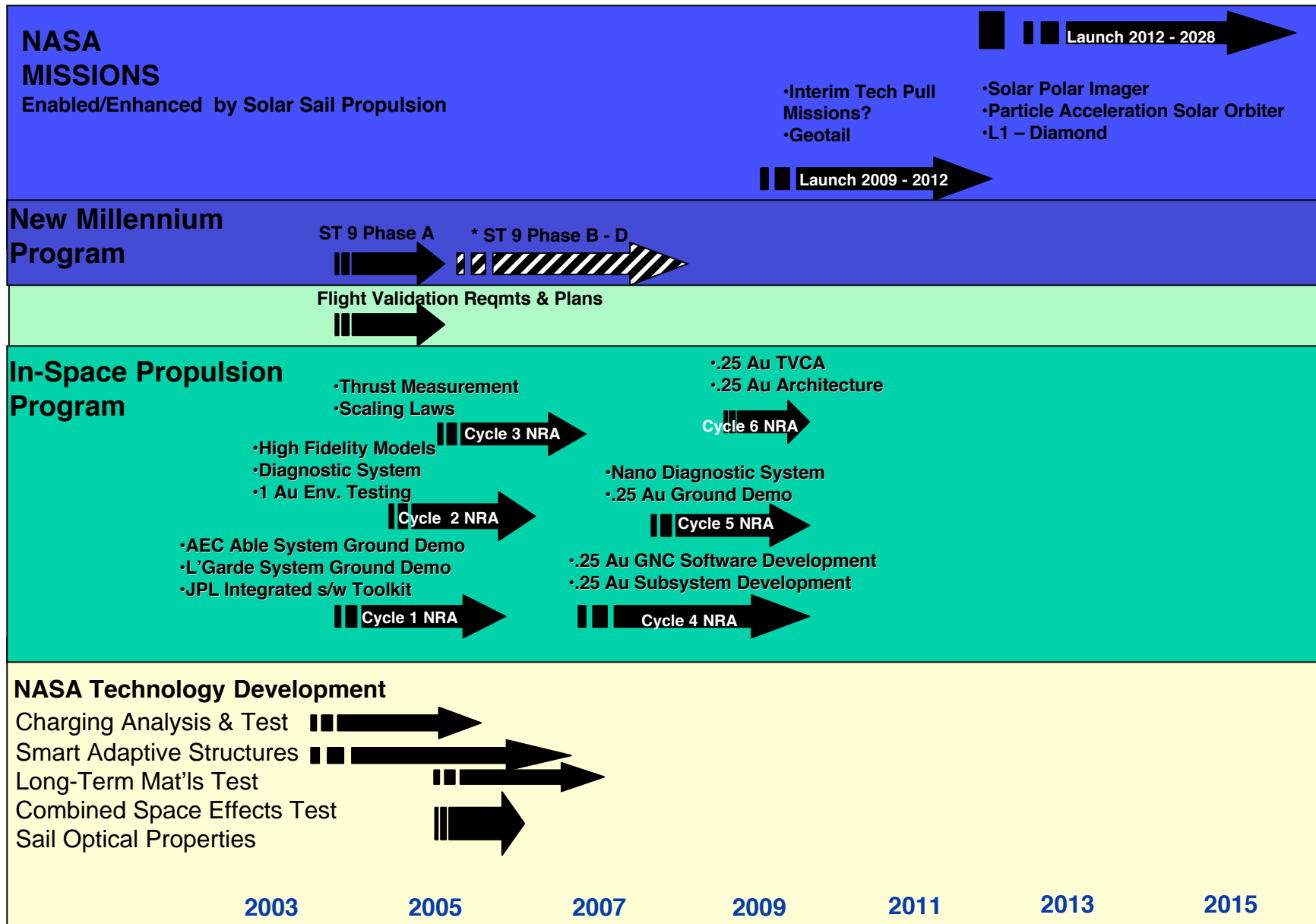


Solar Sail Propulsion Science Mission Work Breakdown Structure





Solar Sail Propulsion Roadmap





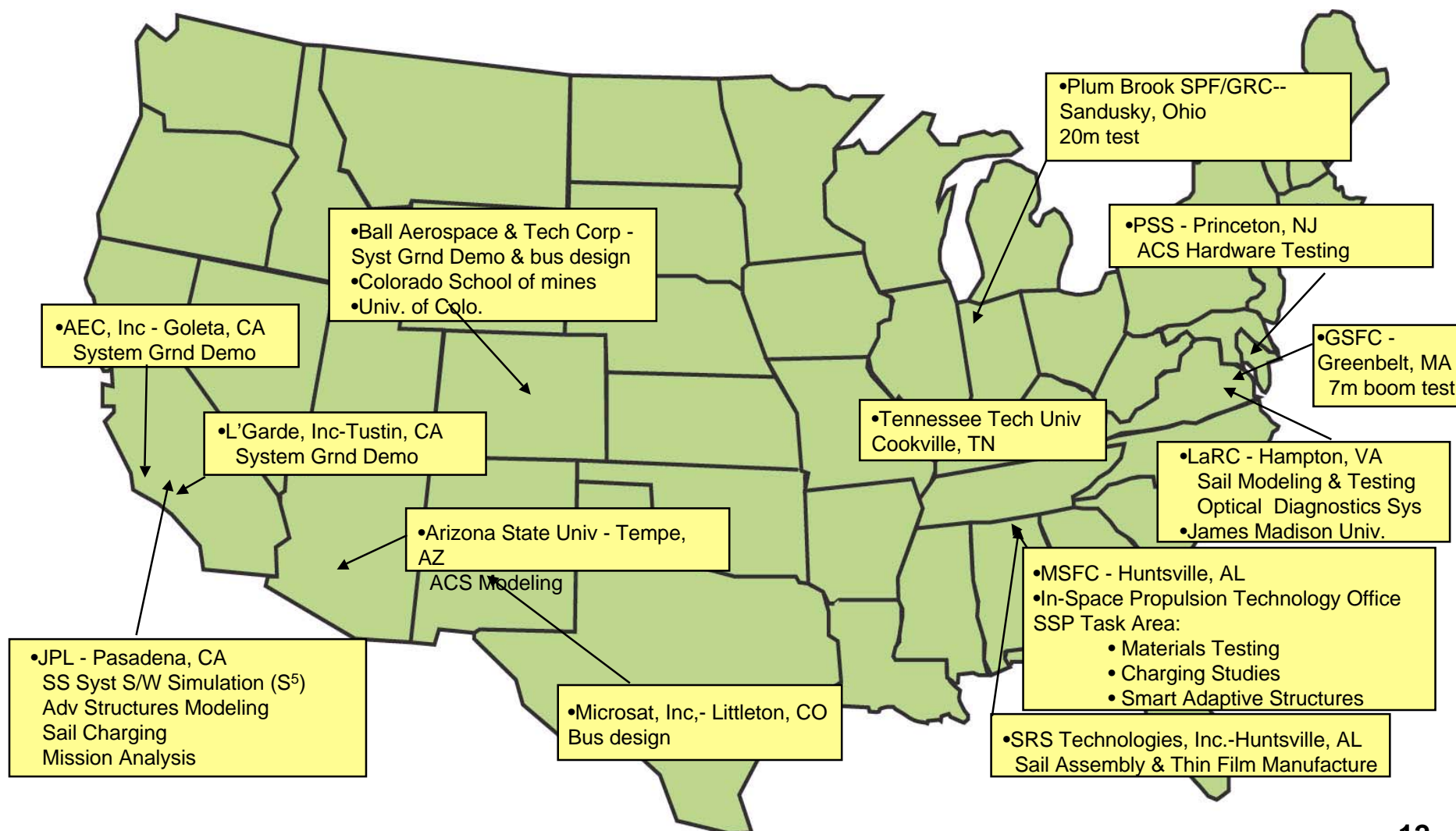
SSP Tasks Funded to Date



- ◆ **NRA Cycles 1 & 2**
- ◆ **Directed Tasks**
- ◆ **Systems Analysis Tasks**
- ◆ **Composite Schedule**



Solar Sail Propulsion Participants





Funded Competed Work



Cycle 1

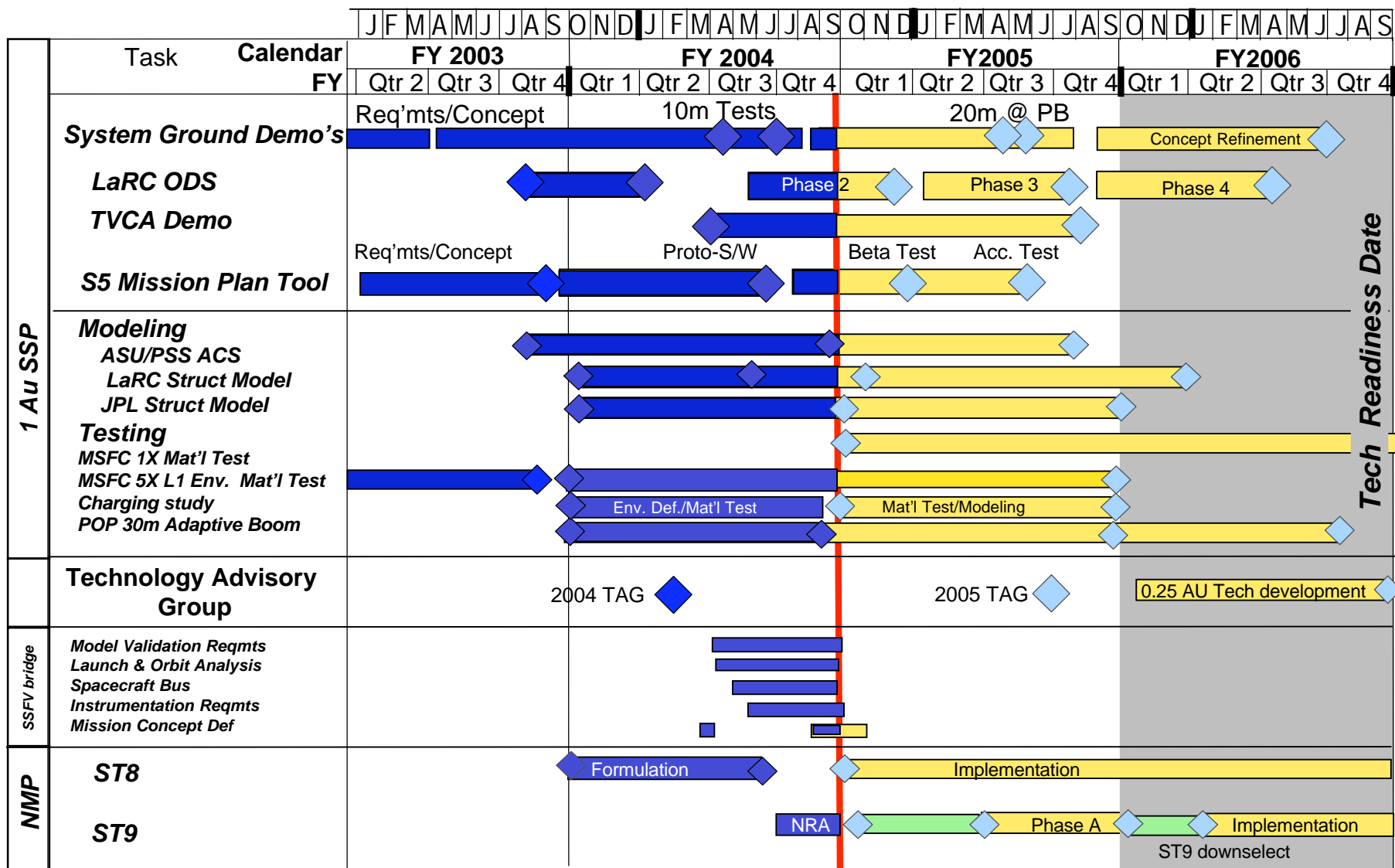
Selection title	Prime
Scalable Solar Sail System Development and Ground Demonstration	AEC
Ground System Demonstrator	L'Garde
Solar Sails GN&C Tool Kit	JPL

Cycle 2

Selection title	Prime
Optical Diagnostics System for Solar Sails	NASA LaRC
Advanced Computational Models and Software for Design and Simulation of Solar Sails Including Experimental Validation	NASA LaRC
Development of a Low-Cost, Low-Mass, Low-Volume, and Low-Power Attitude Determination and Control System (L4-ADCS) and High-Fidelity Computational Models of Solar Sail Systems	Arizona St Univ
Laboratory Characterization of Candidate Solar Sail Material	NASA MSFC
Advanced Manufacturing Technologies for Solar Sails using Processes Developed Specifically for Production of Ultra-thin Solar Sail Materials for Near, Mid and Far Term Space Science Missions	SRS Tech
Structural Analysis & Synthesis Tools for Solar Sails	NASA JPL



SOLAR SAIL PROPULSION ISP & NMP ST9 Major Milestone



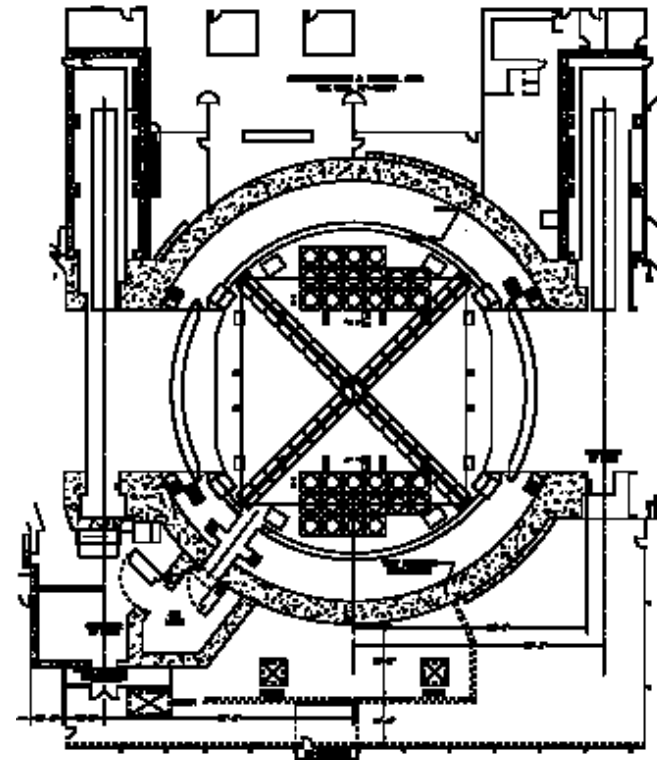
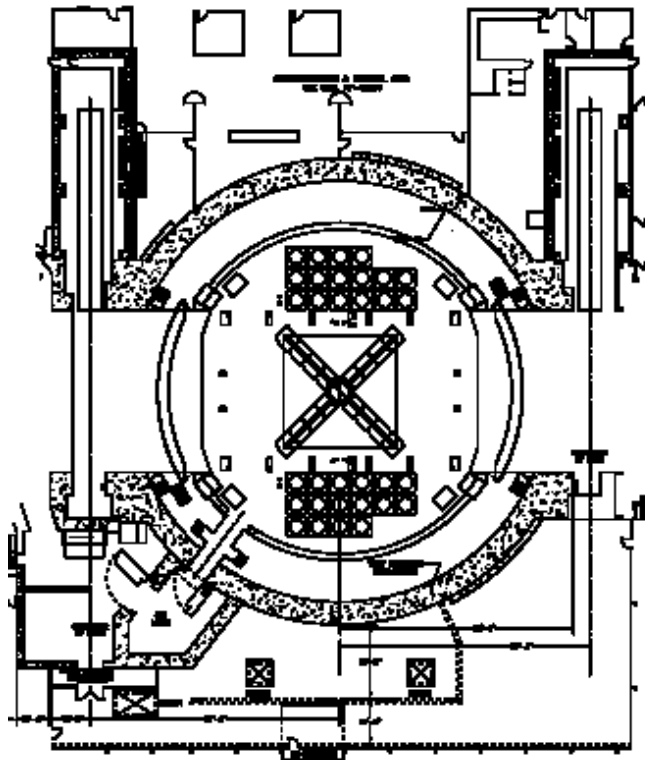


System Ground Demos: Plum Brook 10m & 20m Layouts



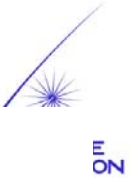
*NASA Glenn Research Center,
Plum Brook Station, Sandusky,
Ohio*

*Test chamber: 30.5m (100ft)
diameter by 37.2m (122ft) high*





SGDs: 10 meter tests



Project Office Level 1
Test Requirements

AEC

L'Garde

Fabrication of scalable test article	No	Designs for 20m require features not necessary just for scaling. 20m booms require very different tip assy and deployment mechs. Sail will have ripstop and corner jumper, 10m did not. 5 micron sail was tested, 3 micron to be used. Patchwork fab/assy method to be used in phase 3 was not used in phase 2. Central structure to change significantly. Trolley needed for gravity offload.	No	Designs for 20m require features not necessary just for scaling. Some refinements from 10 meter design needed in mandrel, gravity off-load higher off ground, line management, manufacture process, addition of a control vane to one boom (cooling, offload, viewing. Etc.), offload structure (more support form above, higher off ground, smaller available footpring at end of beams, boom and sail assembly process. Wrong material in UV evaluation, thermal pipe height and length must change (and not impact test instrumentation)
Demonstrate transportation and handling of test articles and GSE	Yes	After transport from ABLE, test articles and GSE set-up & used at LaRC. No damage identified.	Yes	Some wear showed on a line due to vibrations rubbing Kevlar line during transport
Demonstration of scalable deployment	Part	Unexpected tear in sail in first ambient deployment attributed to sequencer. Border cord might have prevented further propagation of damage that would have affected propulsion performance or structural integrity, but test was stopped and fixed by human intervention. Later deployment was successful	Part	Runaway boom segment inflation and QC failure in packing lines caused half meter tear and several line failures, but not mission critical. Deployment not stopped and did continue to completion without human intervention. No successful repeat possible.
Acquire adequate measurements for the verification of computational models from test data	Yes	Measured gravity sag ~30% smaller than predicted. Although, the third mode shape could not be excited and a slight tweak of model to distribute spreader bar mass was required, multiple modeling results showed unusual high degree of comparison between prediction and test for deflections and mode shapes. Further analysis would be valuable.	Part	Boom tip cap gauges, accelerometers, videogrammetry failed. Low fidelity static deflection data from photogrammetry and dynamic data from laser vibrometer. Measured gravity sag ~30% smaller than predicted. Insufficient analysis of test data. Model frequencies close to prediction.
Mechanical assy drawings and electrical system schematic drawing for the test article and GSE	Part	Drawings made "available", but only partial set uploaded to sTIN. No drawing tree provided.	N	Drawings made "available", but none delivered to STIN. No drawing tree provided.

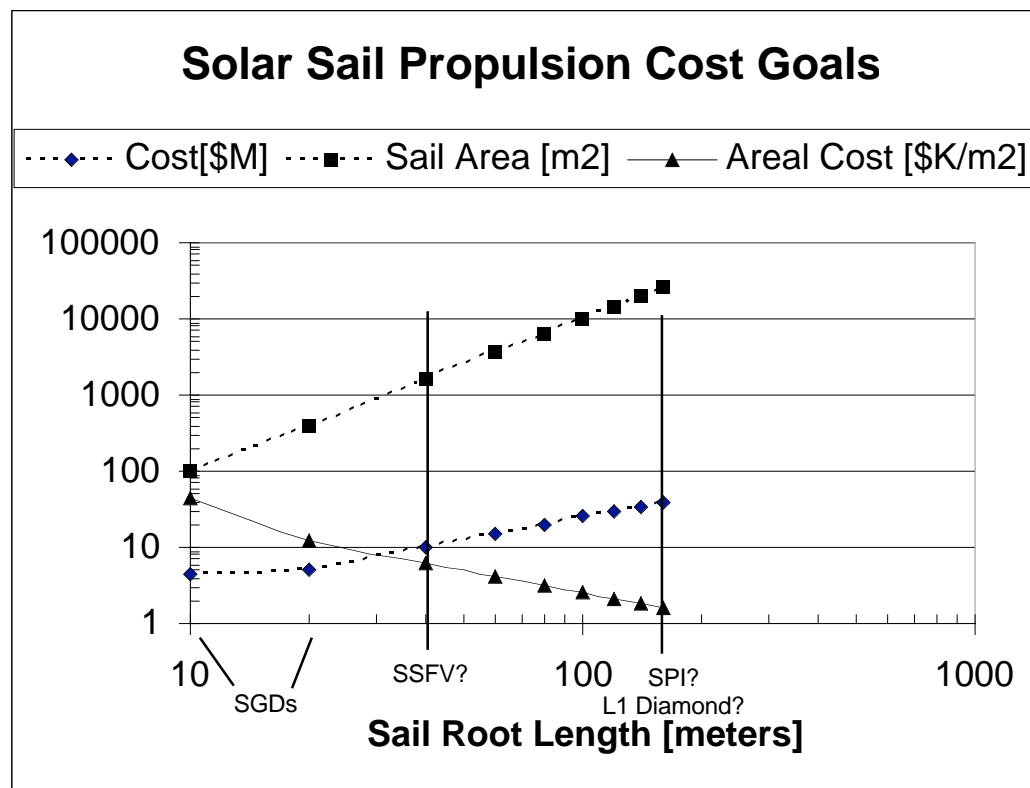
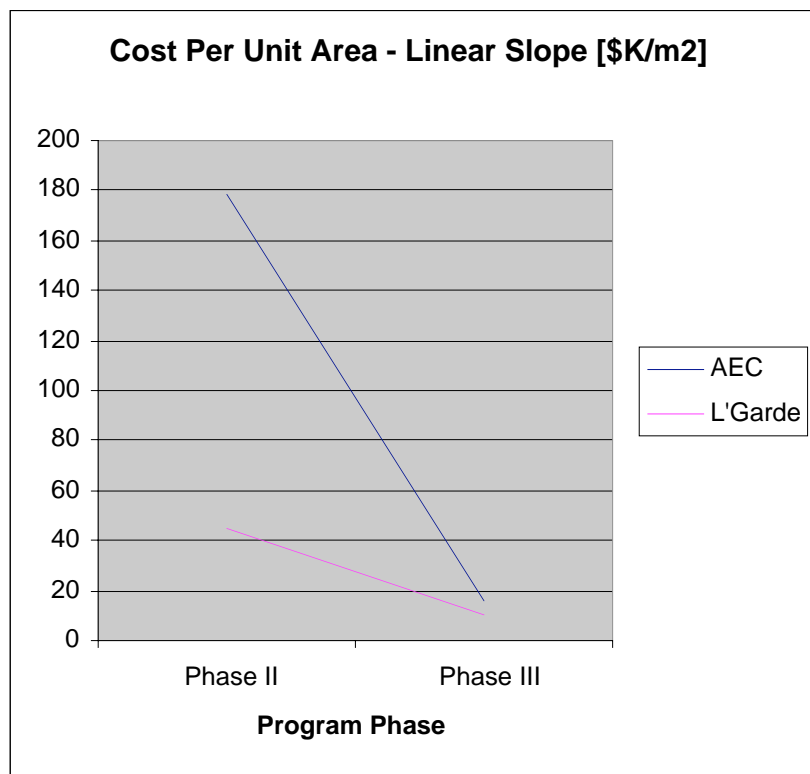


Technology Cost Extrapolation



Current History – Phase II costs indicate
1 vs 4 quadrant 10 meter demonstrators

Areal Costs must continue to reduce to keep
total cost for science missions affordable

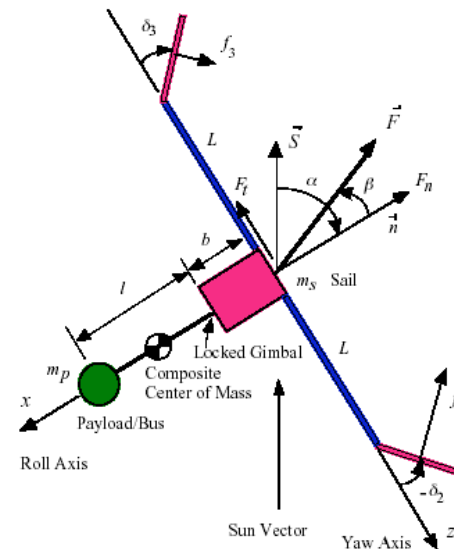
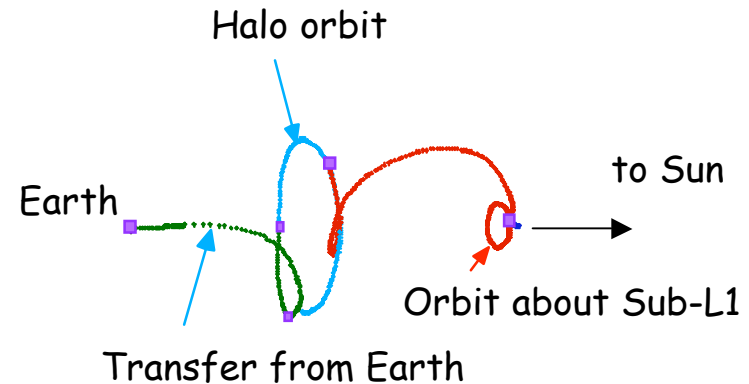


Note: Cost history is relevant to ISP-SSP system ground demonstrators

Solar Sail GN&C Challenges

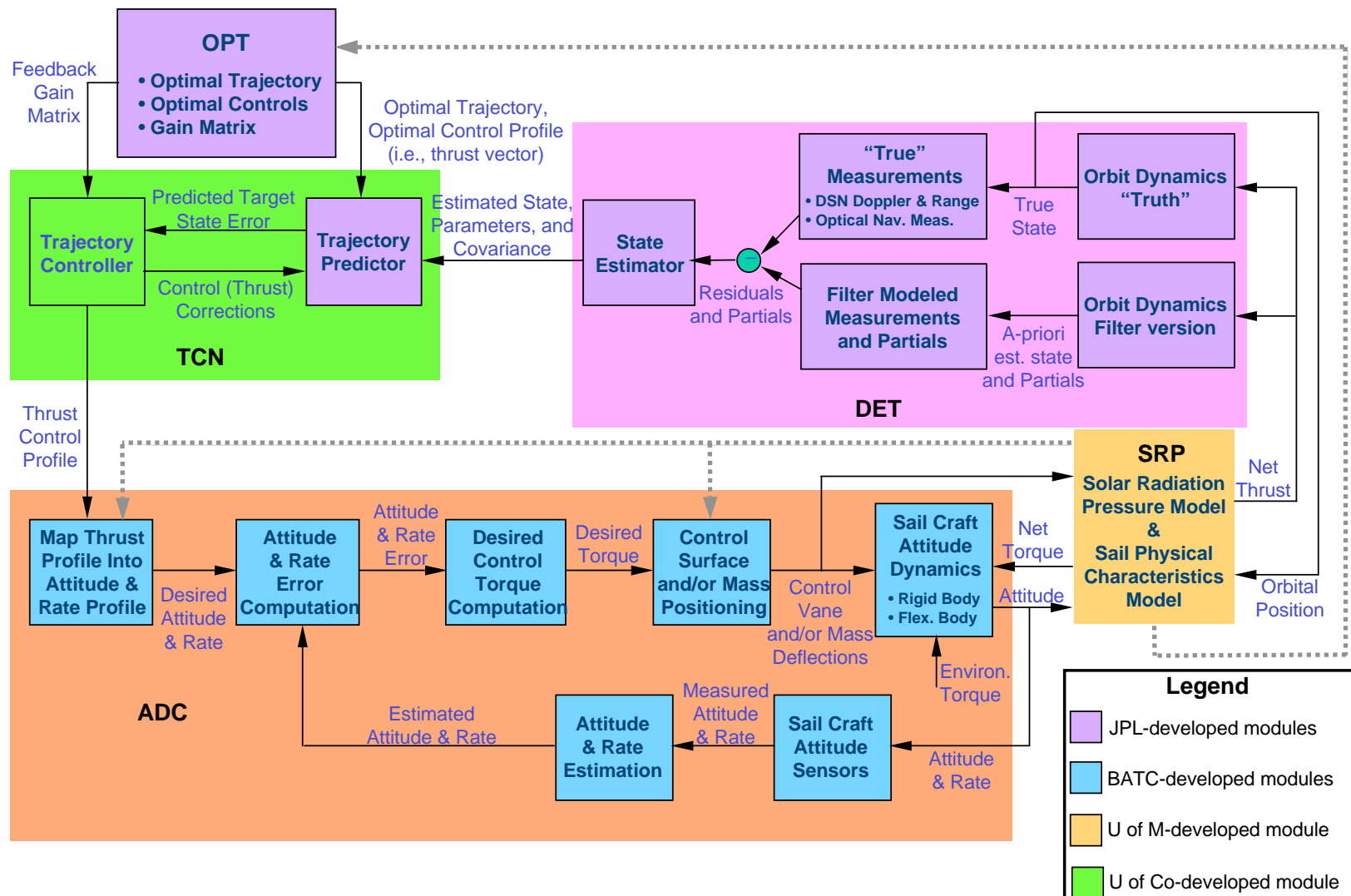
- Low Thrust Trajectories
 - Optimal Control Theory
 - Solar Sails Unique
 - Thrust Vector Constraint

- Attitude Control Systems
 - Optical
 - CP/CM
 - Conventional





S5 GN&C Functional Architecture





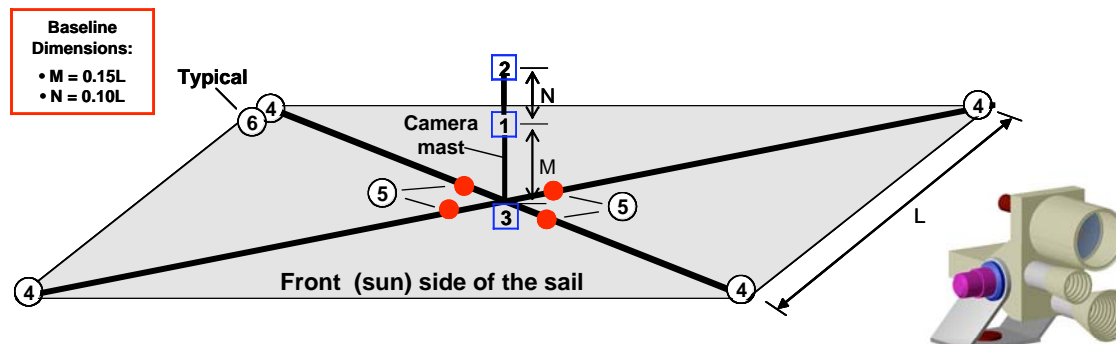
Optical Diagnostic System Task Summary



◆ Award for **Cycle 2 Topic: Integrated Solar Sail Diagnostics Package development**

◆ **PI: Richard Pappa, LaRC**

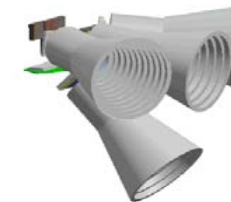
- Ecliptic Enterprises Corporation (Pasadena, CA) – data and space requirements
- South Dakota School of Mines – structural models and analysis
- Texas A & M University – Image processing techniques
- James Madison University – Infrared thermography and fiber optic sensors
- AEC – System Ground Demo Integration
- L'Garde System Ground Demo Integration



- 1 4 front-side 90-deg FOV camera clusters (16 cameras)
- 2 1 front-side inspection camera cluster with pan-tilt (3 c)
- 3 4 cameras at hub, one looking down each boom (4 can
- 4 3 accelerometers at each boom tip (x & y bending, tors
- 5 Boom root strain sensors
- 6 Membrane tension sensors

Also:

- Infrared thermography cameras on back side mast?
- Piezoelectric actuators for dynamic excitation?



Sail and Boom Cameras:
The Cluster Concept



TRL 6 Definition



- ◆ **Origins**
- ◆ **Current Approach**
- ◆ **Potential Impacts**

**TRL 6: Demonstration of a
system in a
relevant environment (ground or space)**




What is the Distinction?



TRL 5: *Component and/or breadboard validated in a relevant environment*


TRL 6: *System/subsystem model or prototype demonstration in a relevant environment on the ground or in space.*













TRL 7: *System prototype demonstrated in a space environment~*



	Subject	Action	Condition
TRL 5	Component and/or breadboard	validated	a relevant environment
TRL 6	System/subsystem model or prototype	demonstration demonstrated	a relevant environment on on the ground or in space. a space environment

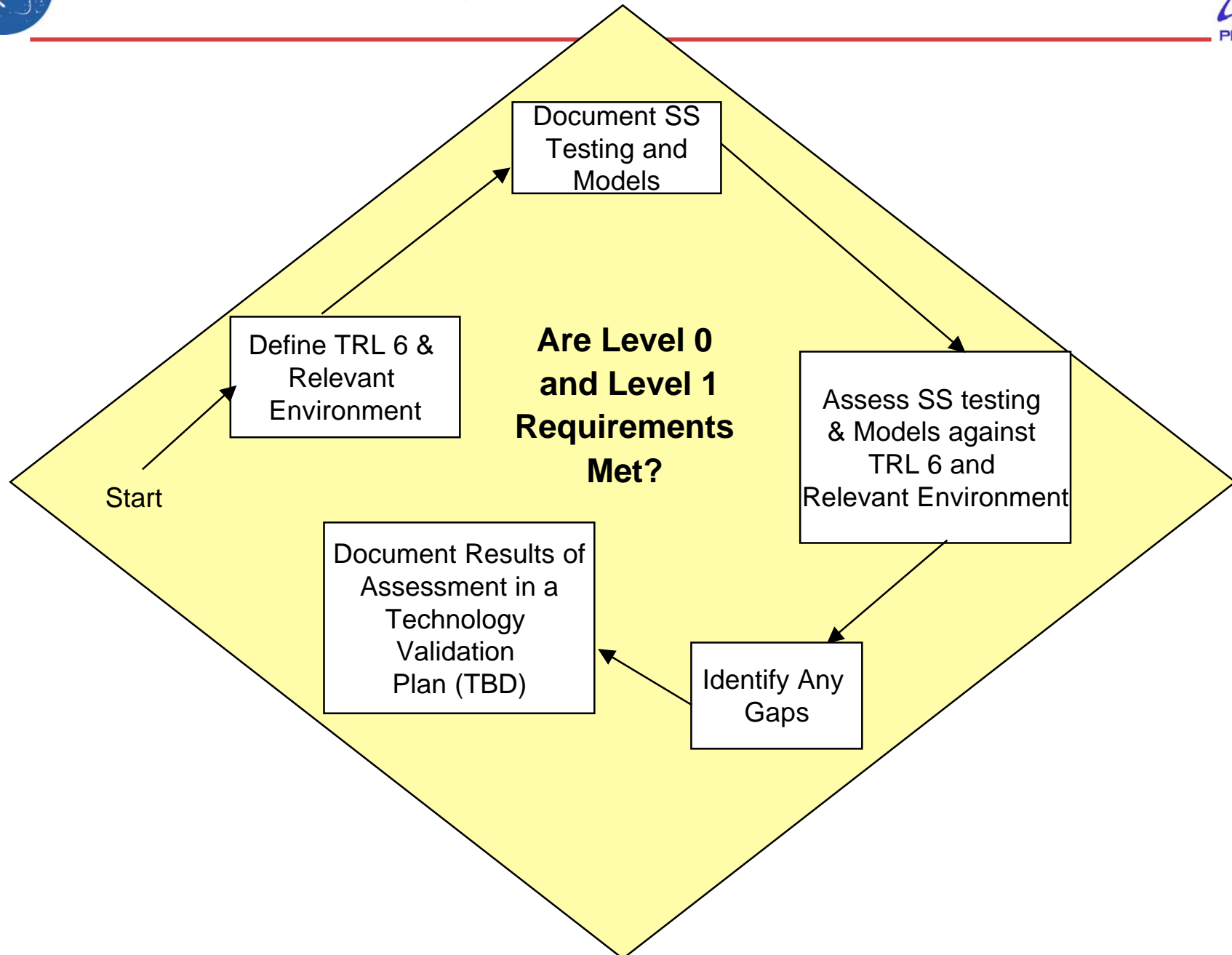
TRL
7



	Component	Subsystem	System	Breadboard	Model	Prototype	relevant ground	Space
TRL 5								
TRL 6								
TRL 7								



Technology Assessment Approach





Future Plans



◆ Tech Development

- NRA Cycles 3a & 4
- FY05 Directed Tasks
- Systems Analysis

◆ Flight Validation

- NMP – ST9

◆ Science Mission

- SPI
- L1 Diamond
- Interstellar Probe
- ??

◆ ROSS NRA Cycle 3a

- Thrust Measurement
- Scaling Laws

◆ ROSS NRA Cycle 4

- TRL 6 Gaps

◆ FY05 Directed Tasks

- Image Rendering for photogrammetry
- Combined Env. Effects
- Long Duration Matl'd Tetst

◆ Systems Analysis

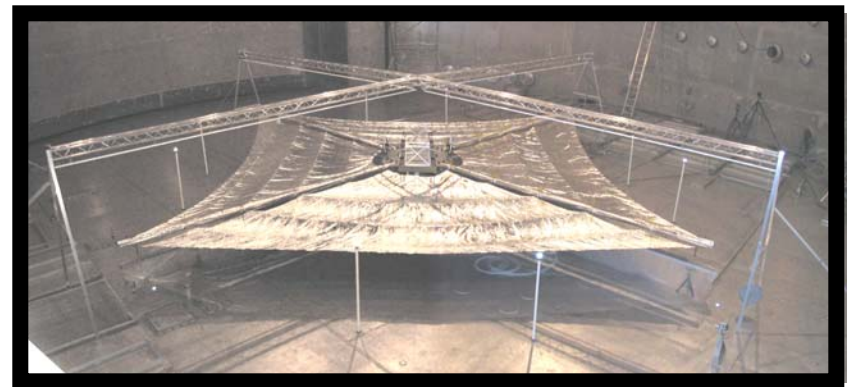
- SPI Vision Mission Study
- L1 Diamond
- PASO



SSP Near Term Challenges



- SSP demonstrations in simulated environments on the ground evolve concepts & capabilities well, but fall short of retiring risk for science roadmap mission applications.
 - full systems too large for conventional test methods/facilities
 - relevant environments ($<10^{-12}g$, 4K, 10^{-6} Torr) available only in space
 - full system thermal vacuum ground tests are a significant investment.
- Current SOTA is maturing rapidly
 - Mechanism designs evolving
 - Fabrication methods evolving
 - Integration with spacecraft(bus) & LV needs continued attention
 - Materials testing and charging studies so far have identified no showstoppers. Design specific issues TBD.
 - System dynamics simulation & analysis beginning
 - S5 useful immediately for trajectory design – unexpected optimization may redefine sail design reqmts.
 - Few unique component types – flight qual at assemblies/subsystem levels





Solar Sail Talking Points



- ◆ **There are missions only solar sails can do--because you can't load up enough chemical propellant to do them conventionally.**
- ◆ **Solar Sails are environmentally friendly, safe,--and no toxic fuels or combustion byproducts, no high temperatures, or electrical currents, relatively small in size and weight.**
- ◆ **There is no doubt it will work--propulsion from sunlight saved a Mariner mission to Venus several decades ago. Recent advances in thin films, composites, and our understanding of how to engineer gossamer structures have made it even more practical and higher performing.**
- ◆ **No super-lasers or acres of unobtainium needed!**
 - Sails large enough to carry people or go to destinations beyond our solar system are still in the future.
 - But there is a current need for science probes and instruments that can be done in the next very few years almost within the current SOTA.
- ◆ **Solar Sail Propulsion (SSP) is the In-Space Propulsion technology activity with a goal of preparing a revolutionary alternative propulsion system for inner solar system science missions through a program of ground development/demonstration then flight validation.**
- ◆ **Sails for radiation hazard sensing and communications platforms may be beneficial to new agency initiatives**



Backups



How SSP Develops Content



◆ Technology Assessment Groups (TAG)

- Objectives
 - To allow technologists and system experts an opportunity to participate in an open forum to exchange information, express ideas, and provide technical insight to those responsible for developing the In-Space Propulsion technologies
 - To identify and address challenges and issues that must be overcome in order to satisfy performance, schedule and system goals to advance the technology
 - To create a technology based roadmap which captures the expertise and recommendations of the TAG participants for consideration into the project management roadmaps
- Approach
 - Identify the system/subsystem level components
 - Examine and determine the technology readiness level (TRL) of the current state-of-the-art technology of each subsystem
 - Access to Technical Papers and Conference Proceedings
 - Availability of unpublished test results from reputable sources
 - Participants' insight into and/or direct involvement with related research, systems analysis, and testing
 - Assess the level of difficulty to advance the technology
 - Identify technology challenges (gaps)
 - Identify potential gap fillers

◆ Develop Roadmap/Investment Plan and Monitor

- TAM & LSE develop/revise roadmap based on
 - Program Leadership Guidance (based on budget, Tech Prioritization, executive decisions)
 - Budget changes
 - Technology maturation
 - TAG inputs
- Reviewed monthly in HQ/FPR and MSFC/PMMR processes
- Reviewed annually in POP exercises
- Presented in Outreach Presentations (NOAA, Planetary Society, Workshops, Conferences)





Solar Sail Propulsion

Internal Management Structure



Edward Montgomery (TD05)
Technology Area Manager
Solar Sail Propulsion

Roy Young (SD20)
Lead Systems Engineer

- DRDs
- TAGs
- Formal Reviews

Joan Presson (TD05)
Assistant TAM

- Budget
- Schedule
- Directed Tasks
- Code U Interface
- SBIR/STTR Interface

Greg Garbe (TD05)
Solar Sail Flt Validation
& Mission Outreach

- NMP ST9
- SPI Vision
Mission Study

Donna Patterson (Sverdrup)
Budget Analyst

- Budget Analysis
- Status Reports
- Funds Control

Tammie Sallo (Gray R.)
Project Analyst

- Monthly Status Rqmts
- Documentation Control
- Facilitator of Formal Meetings & Reviews
- STIN Management

Roy Young (SD20), COTR
David Vaughan (JPL)
Charlie Adams, Sys Eng (Gray R.)
L'Garde GSD Contract

- LaRC
- JPL
- MSFC
- GSFC
- GRC- Plum Brook

Al English (SD20), COTR
David Vaughan (JPL)
Charlie Adams, Sys Eng (Gray R.)
AEC GSD Contract

- LaRC
- Ltwt Attitude Control
Sys, ASU
- MSFC
- Adv Mfg Technologies, SRS
- GRC-Solar Thermal
& Plum Brook

John Rakoczy (SD71)
ODS

Andy Heaton
S5 Mission
Sim Tools

Joan Presson (TD05)
Materials Research

- Modeling & Exp Validation
of SS Charging, MSFC & JPL
- Life Test of SS Material in
Simulated Space, MSFC
- Combined Space Environmental
Effects, GRC & MSFC
- Optical Properties, MSFC & GRC
- Char of Material Including Sail
Joining & Repair

John Lassiter (ED27)
Structures

- Advanced Computational
Methods, LaRC
- Structural Analysis & Synthesis
Tools, JPL & MSFC
- Smart Adaptive Structures,
MSFC